

DECEMBER 11, 1922

AVIATION

VOL. XIII. NO. 24

Member of the Audit Bureau of Circulations

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THE GARDNER, MOFFAT COMPANY, Inc., Publishers

HIGHLAND, N. Y.

225 FOURTH AVENUE, NEW YORK

Subscription price: Four dollars per year. Single copies ten cents. Canada, five dollars. Foreign, six dollars a year. Copyright 1922, by the Gardner, Moffat Company, Inc.

Issued every Monday. Forms close ten days previously. Entered as second-class matter Nov. 22, 1920, at the Post Office at Highland, N. Y., under act of March 3, 1879.

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AVIATION

No. XIII

DECEMBER 13, 1935

No. 58

Lessons of the Grand Prix de Paris

ALTHOUGH the French air transport competition for the Grand Prix de Paris failed to reveal any revolutionary departure in the design of commercial airplanes, it should afford much food for thought to those concerned with the building of air transport.

To begin with, it should be now be obvious that a competition between aircraft embodying new and untried ideas should be held at a time of the year when reasonably good flying weather may be expected. It is true that commercial aircraft will have to fly in almost any kind of weather, yet there are too many important things at stake in a public competition to want to complicate matters by introducing wind and sea or other atmospheric obstacles.

Another point well worth notice in the Paris competition was the provisions made with a view to improving the passenger accommodations, specifying seat dimensions, leg room, etc. But we should like to see a minimum limit placed on the disposable volume of air per passenger, for this factor has an important bearing on comfort, and so does the problem of sufficient ventilation, the solution of which should be especially important.

Perhaps the most significant feature of the competition was the remarkably good showing made by Belgoot Levacovich. This ship embodies the central engine arrangement with which the Germans experimented during the war without, however, reaching a satisfactory solution. For over two years the power plant of the Levacovich underwent exhaustive tests, and to many observers it seemed as if the possibilities of connecting four engines to a single propeller shaft were a hopeless one. Now this ship has successfully passed such severe tests as descending one power and in flight and climbing in this manner, of again starting engine in flight, and last but not least of keeping the whole power plant at 90% horsepower making in flight for several hours. This is a performance which augurs well for the practical realization of the central engine room on aircraft, even though much remains to be done to make the system practical.

Personal "Inventions"

DEFINITE probably to the growing familiarity of the airplane, it seems to invite more unbridled whimsies of the merriest type than formerly. The lighter-than-air field however is still a hotbed for the purely speculative inventor. It is inexcusable also how many inventions come up a new idea simply in terms of "possibility," without referring it to anyone who can make a detailed and impartial study of the subject in comparison with other known methods.

It is not desired here to discourage new or even radical ideas in aircraft design. But it is only fair to put people on their guard against a certain class of ideas which are neither new nor practical, "inventions" which have appeared dozens of times in various forms, but which have always gone on

the rocks when put to the test of quantitative analysis or practical trial.

The following devices for airships, while conceivable, are nevertheless of proven inefficiency compared with other known means, and are based mainly on misconceptions of the conditions involved.

1. Heavy propellers.
2. Compressed air or gas as a substitute for ballast.
3. Inflation with hot air.
4. Means for vertical control on an even keel.
5. "Pancake" hull shapes with no approach to a circular cross section.
6. "Vacuum airship" (absolutely unworkable with present materials).

The following are usually designed to relieve conditions which are nonexistent or nonexistent:

7. Pipes to relieve "motion" behind the hull.
8. Various elaborate schemes to eliminate ballast.
9. Means of increasing the "lift" of the gas.
10. Mechanical substitutes for the screw propeller.
11. Placing the line of thrust at the center of resistance.
12. Means for relieving the force of a "side wind" when in flight.

The following bring in difficulties, kith and kin of all proportion to the advantage gained:

13. Combined airship and airplane.
14. Double or "catamaran" airships.

Inventors and aviation buffs would do well to keep this list on their desk for handy reference before letting their imaginations get out of bounds over "inventions" which are "bound to revolutionize the art of aerial navigation."

Using Obsolete Aircraft

THE occasional reports which the daily press publishes on the progress of the New York to Brazil flight indicate a series of mishaps which have greatly delayed the trip. This is highly regrettable, for these delays have a much greater significance than merely a slow voyage.

The use of obsolete types of aircraft for long flights into unknown regions or into countries whose aeronautical resources are few and far between is obviously unfortunate. The first requirement which any power aircraft should fulfill is ruggedness, for the abnormal conditions under which they are called to operate demand every such of it. Another essential requirement for ensuring the success of such an expedition is the carrying of sufficient spare parts for making minor repairs en route, unless they can be stored at convenient points en route. Failure to carry a spare propeller alone delayed the Bessie Coleman five weeks.

Unless these two fundamental requirements are given the thought they deserve, such trips cannot be successful and American aircraft will suffer from unfavorable comment in the very places where it is intended to land them.

The Grand Prize of Paris

Farman F90 Wins French Air Transport Competition

The French air transport competition for the 100,000 francs Grand Prize of Paris, which was held in the middle of November at Le Bourget airport, was won by Louis Béchereau on a Farman F90 passenger cabin biplane equipped with a 200 hp. Hispano engine. This is the only aircraft that won the Grand Prize, the result of which particularly favored multi-engine aircraft.

The Conditions

The conditions of the Grand Prize of Paris were briefly as follows: The competition was open to French airplanes carrying two pilot's seats in the forward cockpit, and five passenger seats in the rear. The passenger cabin had to be at least 17 ft. wide, four feet high, and make a mean of 6 ft. tall to sit comfortably without leaning. The presence of a crew of two was obligatory on board during all trials. The economical load was calculated at 200 kg. per passenger, of which 20 kg. were to represent the baggage. This load could be represented by sand bags of the weight equivalent to each passenger, in which case, each bag was to be attached on the respective seat, while representing the weight representing the baggage were to be loaded into a special baggage compartment.

The competitors submitted preliminary trials covering general features thought to be desirable in commercial airplanes, and of flying trials which were to bring out safety features. For all trials points were awarded, as shown below: The preliminary trials were divided as follows: (A) basic power performance per passenger carried, 150, 50 and 25 points in order of precedence; (B) Use of radio telephone during flight up to a distance of at least 10 kilometers, 25 points; (C) Stopping and starting one engine of the power plant in flight, 25 points; (D) Stopping the power plant on the ground, the entire crew being on board, 25 points.

The Flying Trials

The flying trials consisted of a safety test, a starting and landing test, and a speed test. In the safety test (E) the contestant had to fly for 15½ hr. carrying full load and to 1 hr. for each wing engine stopped and started, maintaining performance; (F) take off, 15 minutes to a height not exceeding 500 meters and release three 10 m. descending one right handed and one left handed circle; (G) climb after completion of test (F) another 500 meters in 20 min. maintaining performance; (H) climb to a height not exceeding 1500 meters, the horizontal flight for 50 minutes, and the second climb for 1500 meters. The starting and landing tests (I) were determined by the length of the runway. The 25 and 25 points being given for first, second and third places. The speed test (J) was to be run on two consecutive days after a 100 kilometer arrival, as follows: on the first day a distance of 200 kilometers was covered with full load and one obligatory stop made en route, when the power plant had to be stopped, on the second day 400 kilometers had to be covered with one half the specified load. Both trials had to be accomplished without any unscheduled stop. The trial that specified for the 200 kilometer flight, any forced landing

severely the contestant to be disqualified. An average speed of less than 200 kilometers per hour also disqualified the contestant. In these speed tests 3 points were given for every kilometer above the minimum speed up to a total of 200 points, and no additional points were given above 25 and 25 points were awarded for first, second and third place.

The Competition

The great severity of these conditions, which was much criticized by the French aeronautical press, was the unknown which determined the award of points, was undoubtedly responsible for the small number of entries received. The principal characteristics of the entries are given in the accompanying table. The Morane machine of Yvelin "Duc d'Orléans" built under license in France. The Heligol "Lecrother" is a tractor biplane equipped with a central engine room in which four five-horsepower engines are connected by a common-distributor drive, which was described in detail in AVIATION, whereby a landing power unit is automatically thrown out of action if the engine revolution drop below a certain number. The power plant of the Lecrother has two winged tanks for almost two years, and it was recently reported that it had during one trial turned for 2 hr. with a total power output of 600 hp.

Although the Lecrother failed to finish in the competition owing to accessory trouble, the fact that the machine successfully passed the majority of the seven tests is a valuable indication of the possibilities of the central engine room arrangement. As a matter of fact the classification of big interesting trial is not to have come about in an odd way. While flying the speed test, a spark plug broke, whereupon the mechanic disconnected the engine to fit a new plug. But in his hurry to do so he accidentally turned speed the dash lever and so disconnected a second engine. As the pilot was flying at a height of only 150 ft. owing to poor visibility, the loss of 50 per cent of the horsepower caused him to land where he automatically landed him from the competition.

The Caudron C74, piloted by Pando, was a four-engine cabin biplane with two sets of tandem propellers. This ship is a development of the C51 three-engine passenger plane exhibited at the last Paris show. The Farman F90, piloted by Béchereau, represented a refinement of the 770 type which was produced early in 1932 as a combination design to be used as either a four passenger cabin ship or as a cargo observation combat machine. The Potez X is a three-engine cabin biplane, and was exhibited at the last Paris show, in connection with which it was described in the Dec. 16, 1932, issue of AVIATION. The ship failed to arrive in time for the preliminary trials, and therefore did not take part in the competition.

Extremely bad weather began all during the trials, with low visibility, rain and fog, so that the speed tests had to be postponed. The following number of points were awarded in the preliminary trials and in all the flying trials except the speed tests:

Characteristics of the French Transport Airplanes Entered for the Grand Prize of Paris

Model	Engine	Rated hp.	Actual hp.	Gross wt.	Wing span	Max. alt.	Range	Length	Height
Morane	2-270 hp.	540	5274	5070	123.5	125	90.75	50.40	4.05
Lecrother	4-500 hp.	2000	2100	6000	140.0	470	50.00	34.00	5.54
Caudron C74	4-500 hp.	1200	2100	5500	120.0	350*	55.00	50.00	4.87
Farman F90	2-200 hp.	400	1170	5500	95.0	250	32.00	3.20	5.00
Potez X	3-500 hp.	1500	1400	4700	95.0	270	27.00	20.00	4.35

* Estimated.

TABLE SHOWING POINTS AWARDED IN THE COMPETITION

Machine	Power	Alt.	Speed	Stops	Starts	Radio	Total
Farman F90	100	100	100	100	100	100	700
Lecrother	100	100	100	100	100	100	700
Caudron C74	100	100	100	100	100	100	700
Morane	100	100	100	100	100	100	700
Potez X	100	100	100	100	100	100	700

An asterisk (*) indicates that the trial was unsatisfactorily completed.

It does not appear clearly from the regulations of the competition whether failure to comply with all the trials specified constituted a contestant. Judging from the above table, however, and also from the fact that the victorious Farman F90 was a multi-engine aircraft and hence could not carry out test (E)—that is, climb with one engine stopped—it must be assumed the award of the prize rested with the officials of the Aero Club of France.

Owing to the bad weather the first speed trial was postponed to Nov. 14, and it was decided, in order to facilitate the task of the pilots in the low visibility to reduce the length of the course to 40 kilometers. Consequently the course had to be covered each day twice.

Béchereau on the Farman F90 covered the 800 kilometers of the first speed test in 4 hr. 7 min. stopped time, and in 3 hr. 55 min. flying time, which makes an average speed of 180 km/h. or about 100 m.p.h. The Lecrother was eliminated that day as shown stated, and Lachaudon, on the Morane "Gaspard" came down by water near the end of the course. The Caudron C74, piloted by the veteran Pando, crashed on the fifth lap and killed its pilot and two mechanics when the left hand rear propeller broke or flew off its hub and literally cut the fuselage in two, thus depriving the machine of its controls.

The last only the Farman F90 to complete the speed test, which it did the following day by covering the 800 kilometers in 4 hr. 37 min. stopped time, which was an average speed of 153.3 km/h. (95.3 m.p.h.). The 200 hp. Hispano motor 2320 cubic in. a conventional version of the vee type 20 model with which many French commercial airplanes are equipped.

Glider Development

The development of a portable airplane capable of reaching an altitude of 20,000 ft. is the aim of Dr. George H. Madsen, designer of the Hispano glider, who is employed as project engineer at the Glenn L. Martin Co.

Dr. Madsen believes a glider ship should be a height of 20,000 ft. and support not only its own weight, but also that of a pilot seeking to a moving object, such as a battleship or destroyer.

Under these conditions the glider would function as a kite, with the difference that it would be controlled. Each glider would replace kite and observation balloon, and it would have the additional advantage of being less visible to an enemy and so more difficult for its crew to take off attack.

When he wished to descend the pilot would simply drop the cable and glide to a landing place either on the ship or near it.

Dr. Madsen reports of the London Daily Chronicle's observer that German scientists place construction of huge gliders, capable of carrying large loads for hundreds of miles, is described as "hush" by Dr. Madsen.

"Given the same area, or wing surface, a glider must be from five to ten times lighter than a motor-driven plane," he said. "And as the size is increased, the wing loading must be increased to overcome the dead weight of the structure. If the lifting surface is doubled, the dead weight is more than doubled. The larger the plane, the more you approach a point where the plane will not support a useful load."

However, flights of hundreds of miles in a glider are possible. Madsen believes that gliders should be designed to be carried aloft by warm currents, glide on air way to another spot where the currents rise upward, go through the same conditions to gain altitude and so on until darkness or until the destination had been reached.

Thus, of course, would require a sharing of the air currents so pilots would know where to glide to gain altitude.

Glider at night is responsible in the opinion of Dr. Madsen for its use in reconnaissance and observation. He said, "Glider never fly at night."

Italian Type O. S. Airship

This new airship was built lately by the Stabilimento Idroelettico Aerostatico di Ronco, Italy. Her general features do not differ materially from those of the standard type to which Italian aerostats carry over several hundred up to now in the field of warship construction. Her aerodynamic characteristics, however, are a great improvement.



Above: the Italian winged tank O. S. in flight. Below: Close-up view of the O. S.

ment on those of the past, owing to more important improvements effected in the design of the stern and to better cabin stability.

The envelope is 223 ft. long and has a capacity of 160,000 cu. ft. It is made of non-pyric fabric with rubber on both sides. The balloons are separated by means of looped steel partitions.

The tail surfaces are rigidly braced by steel tube struts. These are arranged over the fabric of the envelope and are secured through iron links which serve the same purpose. Along the main spar in question. The tail propeller extends along the lower curvature of the hull from stern to stern. It is built of steel tubes which are joined in such a manner as to render the system practically elastic.

The windows are in completely closed and aerodynamic heads a row of four, six panes. For the convenience of the better comfortable vision are provided. Windows, which can be opened, are fitted on the length of both sides of the cabin, giving a very fine view.

Two 120 hp. Colombo Combi engines are mounted on outriggers at the rear and at the O. S. and drive two wheel propellers of the pusher type.

On her first trial trip the O. S. actually reached a speed of 50 m.p.h. While the O. S. is still an experimental type, its trials have been very successful in indicating that it is a great improvement on other types of airship having the same capacity.

An Aerial Picnic in France

By F. Wallace Kellott

A most interesting illustration of the powerful pleasures of Spain was afforded by the recent semi-picnic given by the Aero Club of France for its members. Tiring of the conventional flying season—which, as told some time ago, in France is four or five different seasons—the directors of the Aero Club decided to give the enthusiasts Spain, and they are successful, something different.

Further to advance the public in aviation, and to encourage aerial touring, they expressed a desire to which everyone should come by air. They first issued the notice "The 194 of the 1944" meaning that 194 invited members of the Club would meet at the 194th kilometer of the Paris-Granville road, at which point lies the charming village of Tilly-sur-Seine. As it happened, however, there were 192 non-members, and the Secretary plans to show the 200th kilometer to accommodate the guests at the next picnic.

As early as nine o'clock in the morning the guests began to arrive. From then until noon the small pasture adjacent to the country inn at Tilly-sur-Seine presented a busy scene as the planes gracefully headed with their engines of aviation. In small numbers and large ones they arrived—there were sport planes carrying one or two people, and great passenger planes carrying a dozen—in all thirty-five machines of almost every type built in France. Last but not least to arrive was M. Laurent Eysa, under-secretary of state for aeronautics, who was warmly greeted by the president, already seated into a happy family.

The good looking pilots, all men, have arms, charming smiles, have flanked with the thrill of the air would go far to what the appetite of any pilot, and every single one had been in Tilly-sur-Seine. At mid-day once the dinner, served at small tables underneath the shade of the "Belle Dole," it was, as old fashioned picnic dinner, delicious, from "appetite to figure." Then came the concert—in which each guest who could sing, or dance or tell a funny story, con-

tributed his or her part. The afternoon passed all too quickly for everyone.

With the ringing of bells, the great mechanical birds, being none more, were put in action again and the "44 of the 1944" was over.



The aerial picnic here shown at Tilly-sur-Seine, just across the border of a cheerful horizon. The French air minister is seated at the left of the picture's left.

took the air to return to their various seats by the light of the setting sun.

In fact not something in events of this kind, taking place with increasing frequency in countries certainly no more progressive than ours, usually of consequence? Here better an aviation, so useful commercially and available as a sport in times of peace, so indispensable in time of war, be brought closer to the heart of the wide-awake patriotic citizen?

New Splittorf Magneto for Aircraft Engines

Ignition System Specially Developed for Aircraft

The Splittorf Electrical Co. has added to its line a new series of magneto known as Models S and SS. This company has long been known by its successful development of the inductor type of magneto. The development of the new series has been necessary for several years and is the direct result of attempting to improve the design of the inductor type magneto as to be increase the efficiency. To obtain a greater electrical output has been the aim in the development of the new series. The new magneto is at present made in two sizes, the smaller one known as Model S and the larger one known as Model SS. The various mentioned in previous articles in *American* describing the winning planes in the Pulitzer and Curtiss-McCormick Trophy Races were the models S and SS magneto.

The Splittorf Model S Magneto is being manufactured in 1, 2, 3, 4, and 6 cylinder types. It is of a very neat design and light in weight.

Constructional Details

The Model SS magneto is similar in construction to the smaller size and is particularly adaptable to larger engines as where a spark of greater voltage is required. Model SS is especially adaptable to aircraft ignition. It is manufactured for engines of from one to twelve cylinders, inclusive.

The following figures taken from the characteristic curves of these magneto prepared by the Engineering Department of the Splittorf Electrical Co. give an idea of the capabilities of the magneto.

The model S is of advance positive pressure spark regularly over a 0.025 in. spark gap under 90 lb. compression at 45 rpm.

In the new Splittorf series the model S has a single permanent magnet, while in the model SS two magnets are used. The magnet or magnets are arranged with the planet traversing to the rotor arm. The effective connection of the magnet to the rotor arm is at right angles.

The design of the rotor is particularly interesting. A single steel shaft supported on ball bearings is used. The rotating laminations are effectively fastened to the shaft by means of being machined in non-magnetic material which is secured around the shaft and laminations. There are several advantages of this construction. A solid shaft naturally makes a more rigid construction than one built up of several parts. The shaft is not magnetized and some magnetic leakage is therefore eliminated. The total magnetic reluctance of the air gap has been reduced by increasing the area.

The frame of the magneto is an aluminum die casting. In four casting are mounted two magnet pole pieces and two ball pole pieces. The casting is made of an alloy especially developed to take care of the difference in shrinkage and temperature. The insulation on which the permanent magnets and the coil core are ground, so are also the ends of the coil core and the coil core. This means solid joints at these points, which further increases the efficiency of the magneto.

The coil core is attached to the frame by means of clamps and the ends of the poles at these points is considerably larger

than in previous designs. More space has been allowed for the coil which permits even greater insulation so that the possibility of electrical breakdown is at a minimum. Primary



Splittorf model SS magneto with the front cover removed.

wires are soldered to the coil terminals, this being done to insure good contact.

The breaker mechanism follows in general Splittorf practice in which the breaker bar and stationary contact point are secured on a plate which is movable for advance and retard. The breaker bar is operated by means of a cam on the rotor shaft. By having the contact points fixed it is possible to ensure them while the magneto is in operation. The bearing of the breaker bar shaft is not required to take thrust, a tapered connection, one end of which is fastened to the breaker bar and the other to the magneto frame conducts the thrust from the movable contact point to ground. The breaker is so constructed as to suit the high operating speeds common with aircraft engines.

The main condenser, which is of standard Splittorf pattern, is mounted on the breaker base and is carried in a metal housing. On the face of the condenser is a coil illustrating the method of adjusting the contact points. On model SS the condenser is attached to the frame of the magneto in the distributor compartment.

The advance lever is mounted in the front cover and by means of a short shaft and crank engages a slot in the breaker base, thereby allowing the base to be moved for advance and retard position.

Assembly

The distributor is mounted above and in the same compartment with the breaker. This compartment is closed by means of the front cover, which when removed provides an excellent means of adjusting the distributor. The distributor disc is attached to the distributor shaft and in the model S magneto is provided with one segment which contains the brushes in the distributor block. In the model SS Advance Magneto the distributor disc has a non-magnetic brass which replaces with segments in the distributor block.

The mounting materials used in the New Splittorf Series are Aluminum and Delrin, the first being an insulating compound of high dielectric and mechanical strength provided by the Splittorf Company.

The cover is formed from a single piece of non-magnetic sheet metal, shaped at its edges to assure waterproofing. The cover is held in place by clamps secured into the frame. These clamps are held by screws which are wired and sealed at the Splittorf factory.

It will be of interest to note that the construction of the cover of the Splittorf magneto is absolutely within the control of the Splittorf Co. in that it contains its own wiring, its own coils—makes its own magnets—makes its own insulating material, etc.

German Aeronautic Emigration

For some time past German aeronautic emigration have been struggling to form a unit, or have been trying to do so, with a view to concentrating the construction of aeroplanes in the north of Germany aircraft construction.

The attempts of the Deutsche und Schachtel-Luft company to form a United States of Germany and France are well known, and it appears from recent indications that the Spanish Republic company has already been formed for the creation of a huge aerodrome at Berlin.

Germany has aviation interests have suddenly been suddenly been enlarged from the perspective of which would not be interrupted by Allied bombing restrictions, while the other enterprise would have an intention to establishing the plans of new "unmanned" aircraft and aeromarine planes. After it others, the German aviation industry has been reported to be contemplating to establish an aircraft factory in Rome. An experimental Munich engine of great power was recently built and tested in Holland, while several other German firms have built new types of aircraft in Sweden.

The latest reference to the marks of Germany's aeronautic emigration is Herr Rohrbach, late chief engineer of the Regensburg-Bomben airplane works, who is built a four-engine all-metal monoplane. He has just entered into a new organization firm at Copenhagen, Denmark. This firm, called the Roforsky, will be owned by Danish managers Erik Roforsky, and controlling several other German aircraft in America, and Herr Rohrbach will be the chief engineer.

As Others See Us

From The Aeroplane (London).—

"All concerned with American aviation desire to be recommended briefly on the engineering results achieved by American airplanes in the Pulitzer Trophy Race and in Mr. Macmillan's speed record attempts therewith. There was a time when we in Europe were justifiably surprised about the progress published by American records, whether fast speed, altitude, distance, or distance, but in those days when the United States Air Service are themselves naturally concerned with all such records it may be taken that figures are accurately and intelligently accurate. In fact, they are probably more accurate than those of any other country in that for altitude and speed performance the most highly developed types of measuring instruments are used.

"Mr. Macmillan's wonderful speed performance more than justified the principle on which the United States Air Service (Military and Naval) recognize the American Aircraft Industry. The record-breaking performance of today is in fact actually the single most factor of knowledge at any time from which the future is best developed. It is therefore very well worth the while of the services in any country to act on the same principle and order the building of pure speed machines."

British Imperial Air Mail Survey

The civil aviation authority of the British air ministry on July 1922 conducted a preliminary complete survey an imperial air mail services and their possibilities. It is proposed to have several services looking up an air line to India, by means of privately operated operations sufficiently indicated to secure a small percentage in capital received.

of ships) Dr. L. B. Tuckman, physicist of the Bureau of Standards, Dr. Max M. Mark, aeronautical expert of the National Advisory Commission and Prof. W. W. Waters, Page of the Civil Engineering Department of Johns Hopkins University, Baltimore, Md. For practically five months were the first meeting in June, which was called to order by Rear Admiral D. W. Taylor, U.S.N., the sessions have been studying and checking the plans of the airplane building in all three full meetings.

Briefly the specifications for the naval airplane call for a single craft built of duralumin, iron and steel, weighing twenty separate gas bags totaling 2,155,200 cu ft, and covered with a single envelope. Her total length is 180 ft and her diameter is 35 ft. Six separate cars are suspended from her hull, each carrying a 300 hp. Packard aircraft engine.

Aircraft Carriers and Naval Defense.—That the air fleet of an army will never get within striking distance of our coast as long as our aircraft carriers are able to carry the proper distance of air power to us in the opinion Rear Admiral William A. Moffat, Chief of the Bureau of Aeronautics, expressed in his return, recently, from an inspection of the new airplane carrier Langley which is carrying on experiments in Chesapeake Bay.

"I have seen planes universally land on, and take off from, the Langley," said Admiral Moffat, "and a thorough test was made of the features of the ship which will enable our airplane carriers to afford the maximum of service in aircraft defense and offense to the Navy. We are working along common lines in the development of the carrier. All developments that are incorporated in the Langley have been evolved by American talent, and represent the most advanced thought and careful investigation on the subject."

Admiral Moffat urged that the full tonnage of airplane carriers allowed this country under the Limitation of Armaments Treaty be built. "We are allowed a tonnage of 10,000 tons to construct for aircraft under the Limitation of Armaments Treaty," added Admiral Moffat, "and we will need every ton of this to supply an adequate aviation complement to the fleet. Other nations have agreed that this figure represent the maximum of our armaments and will in no event afford to ride ourselves lower than the rating which we have been assigned by international agreement."

The Langley, which is the first airplane carrier to be built in this country, has been operating in Chesapeake Bay and off the Virginia Capes since September, but could scarcely be seen in operation as she attempted to carry to the primary work of organization on the new ship. The first landing was made on her deck by Lt. Col. Charles Chandler, U.S.N. The Langley is commanded by Capt. S. H. R. Doyle, U.S.N., who was formerly commanding officer of the Naval Air Station at Hampton Roads.

Officers Selected for Aviation Training.—The following named naval officers have been selected for aviation instruction in the class commencing at Pensacola Naval Air Station Jan. 1, 1933:

Ens. Edgar L. Adams, Ens. Otto Carter Anderson, Ens. Russell D. Bell, Ens. Harlan Lee Brooker, Ens. R. E. Brown, Lieut. James L. Collier, Lieut. Arthur B. Craig, Ens. George Mervyn Doremus, Ens. John F. Ryan, Lieut. Louis H. Fawcett, Ens. Ray E. Farnsworth, Lieut. George H. Fawcett, Ens. Ralph W. Flood, Ens. Oliver W. Galloway, Ens. Hilbert F. Gearing, Ens. Richard P. Olsen, Ens. Marvin H. Green, Lieut. Charles Graham Holgren, Lieut. C. L. Hayden, Lieut. Colman B. Hendon, Ens. Louis Doremus, Lieut. George H. Hendon, Lieut. George H. Hendon, Ens. C. M. Hastings, Ens. Chas. H. Hatcher, Lieut. Conrad A. Kent, Lieut. Scott A. Mace, Ens. Elsworth D. Karkhan, Ens. G. A. Patterson, Ens. John Perry, Lieut. Douglas A. Powell, Ens. Frederick W. Roberts, Lieut. Charles Harry Royster, Lieut. Harry A. Royster, Lieut. J. B. Senter, Ens. William Taylor Smith, Ens. Morton B. Starnes, Lieut. Eugene K. Swenson, Ens. Barnett Thomas Talbot, Ens. Jackson B. Tate, Ens. R. E. Thorne, Ens. C. L. Tyler, Ens. B. P. Ward, Ens. C. E. Woodson, Ens. John P. Wright.

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